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Course details

Course content – Fluids

Navier-Stokes Equations (A)

Continuity equation (A

Mass flux (A)

Boundary Layer Flow (A)

Conservatior of mass (A)

Continuity Equation (A)

SOE3211/2 Thermofluids and Energy Conversion A/B Fluid Dynamics

Dr Gavin Tabor/Dr Christos Makropoulos

Autumn 2006

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and Energy Conversion A/B

Course details

- Course content – Fluids
- Navier-Stokes Equations (A)
- Continuity equation (A
- Mass flux (A)
- Boundary Layer Flow (A)
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- Continuity Equation (A)

Course schedule

11 lectures – Mon 12pm, Fri 12pm Tutorial – Tue 12pm Labs – Wed 11am-1pm

Assessment

- Assessment sheets + labs = 30%
- 1 exam = 70%

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Course content - Fluids

Split into **A** and **B** categories

- A material core B.Eng, M.Eng
- **B** material more advanced M.Eng alone

Assume knowledge of :

- Mathematics partial derivatives, diff. equations
- Fluid dynamics potential flow, Bernoulli, pipe flows, some integral methods

Text :

"Fluid Mechanics", Douglas, Gasiorek, Swaffield

Additional 2 web lectures !!

- Dimensional Analysis (A)
- Blasius solution of b.l. flow (B)

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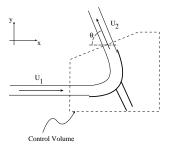
Boundary Layer Flow (A)

Conservation of mass (A)

Continuity Equation (A)

Navier-Stokes Equations (A)

Encapsulate conservation of mass, momentum, (energy). Used these before – e.g. force exerted on vane :



Draw a box around the flow and examine the momentum entering and leaving.

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Navier-Stokes Equations (A)

Continuity equation (A) Mass flux (A Boundary

(A)

Conservation of mass (A)

Continuity Equation (A)

$$\begin{array}{l} \mbox{Change in} \\ \mbox{momentum} \end{array} = \begin{pmatrix} \mbox{Momentum} \\ \mbox{in} \end{pmatrix} - \begin{pmatrix} \mbox{Momentum} \\ \mbox{out} \end{pmatrix} \\ = \mbox{Force on blade} \end{array}$$

x-direction :

$$F_x = \rho A U_1^2 - (-\rho A U_1^2 \cos \theta) = \rho A U_1^2 (1 + \cos \theta)$$

y-direction :

$$F_y =
ho A U_1^2 \sin heta$$

Often referred to as NSE in *integral form*, or *control-volume* formulation.

What happens if the velocity varies across a surface? Split surface into little pieces dA and integrate.

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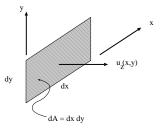
Volumetric flow rate

$$Q = A \times u$$

Easy if u constant – what happens if u varies

$$u = u(x, y)?$$

Need to consider small area dA. Eg cartesian coordinates :



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Course content – Fluids

Navier-Stokes Equations (A

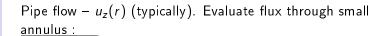
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Area of annulus :

$$dA = \pi (r + dr)^2 - \pi r^2$$
$$= 2\pi r dr$$

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Volumetric flow through dA

$$dQ = dA \times u_z = 2\pi r dr \times u_z$$

Sum this over whole area \equiv integrating

▶ u₂(r)

$$Q = \int_0^R 2\pi r u_z(r) dr$$

Mass flux (A)

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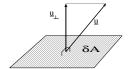
Mass flux (A)

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Conservatior of mass (A)

Continuity Equation (A) Introduce the concept of the *amount* of fluid flowing through a (possibly arbitrary) surface – a flux:

Mass flow through $A = \rho u_{\perp} A = \rho \underline{u} . \underline{A}$



 $\phi^{(\rho)} = \rho u A$ we call the

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(Mass Flux)

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Boundary Layer Flow (A)

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At a wall boundary, the velocity parallel to the wall must be zero :

 $u_{\parallel}=0$

(Actually, more generally $u_{||} = V$). Away from the wall, the velocity is non-zero.

Hence there must be a region of influence of the wall, called the *boundary layer*, where the flow adapts to the presence of the wall.

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Navier-Stokes Equations (A)

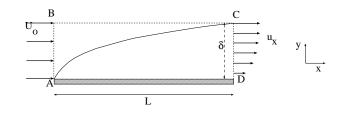
Continuity equation (A

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Across CD,

$$\frac{u_x}{U_0} = \begin{cases} \sin\left(\frac{\pi y}{2\delta}\right) & 0 \le \frac{y}{\delta} \le 1\\ 1 & \frac{y}{\delta} > 1 \end{cases}$$

(a)

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What is the mass flux through BC?

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Mass flux $\phi_{AB}^{(
ho)} =
ho U_0(\delta imes d)$

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Mass flux $\phi_{CD}^{(\rho)}$?

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Mass flux
$$\phi_{AB}^{(
ho)} =
ho U_0 (\delta imes d)$$

Mass flux
$$\phi_{CD}^{(\rho)}$$
?

Small element $\delta A = \delta y \times d$ at y Flux $\phi_{\delta A}^{(\rho)} = \rho u_x dA = \rho u_x (d \times dy)$

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Mass flux
$$\phi_{AB}^{(
ho)} =
ho U_0 (\delta imes d)$$

Mass flux $\phi_{CD}^{(\rho)}$?

 $\begin{array}{ll} \text{Small element } \delta A = \delta y \times d \text{ at } y \\ \text{Flux} \qquad \phi_{\delta A}^{(\rho)} = \rho u_x dA = \rho u_x (d \times dy) \end{array}$

$$\phi_{CD}^{(\rho)} = \int_0^\delta \phi_{dA}^{(\rho)}$$
$$= \rho U_0 d \int_0^\delta \sin\left(\frac{\pi y}{2\delta}\right) dy$$
$$= \rho U_0 d \times \frac{2\delta}{\pi}$$

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Thus :

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Continuity Equation (A) Mass flux through BC = (Flux in) - (Flux out) = $\rho U_0 \delta d \left[1 - \frac{2}{\pi} \right]$

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Mass flux in = Mass flux out

so, over any region V with faces A_i ,

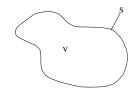
 $\sum_i \phi^{(
ho)}_{A_i} = 0$ — true for *incompressible* fluids

More generally, mass can collect in region V. Mass of fluid density ρ :

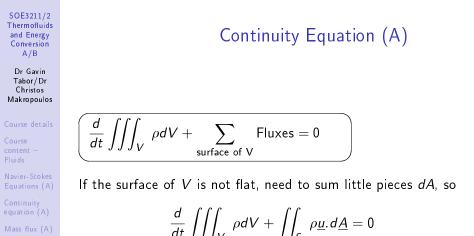
$$\iiint_V
ho dV$$

so change of mass in V is

$$\frac{d}{dt} \iiint_V \rho dV$$



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Boundary Layer Flow (A)

Conservation of mass (A)

Continuity Equation (A)