SOE3211/2 Thermofluids and Energy Conversion $$\rm A/B$$

Fluid Dynamics

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1.1 Course details

Course schedule

11 lectures – Mon 12pm, Fri 12pm Tutorial – Tue 12pm Labs – Wed 11am-1pm

Assessment

- Assessment sheets + labs = 30%
- 1 exam = 70%

1.2 Course content – Fluids

Split into \mathbf{A} and \mathbf{B} categories

- A material core B.Eng, M.Eng
- **B** material more advanced M.Eng alone

Assume knowledge of :

- Mathematics partial derivatives, diff. equations
- Fluid dynamics potential flow, Bernoulli, pipe flows, some integral methods

Text : "Fluid Mechanics", Douglas, Gasiorek, Swaffield

Additional 2 web lectures !!

- Dimensional Analysis (A)
- Blasius solution of b.l. flow (B)

1.3 Navier-Stokes Equations (A)

Encapsulate conservation of mass, momentum, (energy).

Used these before - e.g. force exerted on vane :



Draw a box around the flow and examine the momentum entering and leaving.

$$\begin{array}{l} \text{Change in} \\ \text{momentum} = \begin{pmatrix} \text{Momentum} \\ \text{in} \end{pmatrix} - \begin{pmatrix} \text{Momentum} \\ \text{out} \end{pmatrix} \\ = \text{Force on blade} \end{array}$$

x-direction :

$$F_x = \rho A U_1^2 - (-\rho A U_1^2 \cos \theta) = \rho A U_1^2 (1 + \cos \theta)$$

y-direction :

$$F_y = \rho A U_1^2 \sin \theta$$

Often referred to as NSE in integral form, or control-volume formulation.

What happens if the velocity varies across a surface? Split surface into little pieces dA and integrate.

1.4 Continuity equation (A)

Volumetric flow rate

 $Q = A \times u$

Easy if u constant – what happens if u varies

$$u = u(x, y)?$$

Need to consider small area dA. Eg cartesian coordinates :







Area of annulus :

$$dA = \pi (r + dr)^2 - \pi r^2$$
$$= 2\pi r dr$$

Volumetric flow through dA

$$dQ = dA \times u_z = 2\pi r dr \times u_z$$

Sum this over whole area \equiv integrating

$$Q = \int_0^R 2\pi r u_z(r) dr$$

1.5 Mass flux (A)

Introduce the concept of the *amount* of fluid flowing through a (possibly arbitrary) surface – a flux:

Mass flow through $A = \rho u_{\perp} A = \rho \underline{u} \underline{A}$



(Mass Flux)

Boundary Layer Flow (A)

Boundary Layer Flow (A) 1.6

At a wall boundary, the velocity parallel to the wall must be zero :

$$u_{\parallel} = 0$$

(Actually, more generally $u_{\parallel} = V$). Away from the wall, the velocity is non-zero.

Hence there must be a region of influence of the wall, called the boundary *layer*, where the flow adapts to the presence of the wall.



Across CD,

$$\frac{u_x}{U_0} = \begin{cases} \sin\left(\frac{\pi y}{2\delta}\right) & 0 \le \frac{y}{\delta} \le 1\\ 1 & \frac{y}{\delta} > 1 \end{cases}$$

What is the mass flux through BC? Mass flux $\phi_{AB}^{(\rho)} = \rho U_0(\delta \times d)$

Mass flux $\phi_{CD}^{(\rho)}$?

Small element $\delta A = \delta y \times d$ at y

 $\phi_{\delta A}^{(\rho)} = \rho u_x dA = \rho u_x (d \times dy)$ Flux

$$\begin{split} \phi_{CD}^{(\rho)} &= \int_0^\delta \phi_{dA}^{(\rho)} \\ &= \rho U_0 d \int_0^\delta \sin\left(\frac{\pi y}{2\delta}\right) dy \\ &= \rho U_0 d \times \frac{2\delta}{\pi} \end{split}$$

Thus :

Mass flux through BC = (Flux in) - (Flux out)
=
$$\rho U_0 \delta d \left[1 - \frac{2}{\pi} \right]$$

1.7 Conservation of mass (A)

Mass flux in = Mass flux out

so, over any region V with faces A_i ,

$$\sum_{i} \phi_{A_{i}}^{(
ho)} = 0$$
 — true for *incompressible* fluids

More generally, mass can collect in region V. Mass of fluid density ρ :

$$\iint_{V} \rho dV$$
$$\frac{d}{dt} \iint_{V} \rho dV$$



so change of mass in V is

1.8 Continuity Equation (A)

$$\left(\frac{d}{dt}\iiint_V \rho dV + \sum_{\text{surface of V}} \text{Fluxes} = 0\right)$$

If the surface of V is not flat, need to sum little pieces dA, so

$$\frac{d}{dt} \iiint_V \rho dV + \iint_S \rho \underline{u}. d\underline{A} = 0$$